OPNAY FORM 3750-1A (Rev. 3-63) Page 2

Para 66, OPNAV INSTRUCTION 3750.6, effective ed

	NIFACTURE	2. PLIGHT HAS SINCE ACCEPTANCE	NO. OF	4. HONTHS SINCE LAST PAR/OVERHAUS	S PLT HAS SINCE LAS PAR/CYERY	6 LAST/PAR OVERHAUL AUL ACTIVITY	LAST CHECK PERFORMED	S. FLIGHT HOURS SINCE LAST CHECK	SINCE LAST CHECK
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### THE ACCOUNT

Part V - The Accident

The accident occurred on 31 July 1964. Flight leader Capt.

(Chieftain 2-1), and his wingman 1/Lt. G. F. R. HANKE (Chieftain 2-2) briefed for a scheduled weapons flight. All phases of the mission including NATOPS and emergency procedures

(b) (5)

During engine turn-up, 1/Lt HANKE noted that his fuel boost pressures fluctuated between 28-37 pei. Although the fluctuation is unusual, the pressures were within limitations and all other indications were normal. He checked his fuel panel switches for proper position and accepted the airplane.

Taxi was normal. 1/L't HANKE lined up right as number two man for a section take-off. The flight was cleared by the tower for take-off and frequency switch to departure control.

Both aircraft commenced run-up in preparation for a controlled VFR departure. Observing a thumbs-up from number two, the leader gave brakes release signal, advanced power, checked his wingman in position, and selected minimum afterburner. During take-off roll, throttle was increased to 3/h afterburner. Lift-off was normal.

The leader came out of burner at the field boundary and continued to accelerate at 98%. The number two man was unable to raise his gear and found it necessary to alternate between maximum military and minimum afterburner power to maintain formation integrity. He continued to cycle his gear handle. Accelerating through 210-270 KIAS he requested the lead to reduce power. This transmission was not received by Chieftain 2-1.

At approximately 330 KIAS, the leader noticed his wingmans configuration and transmitted that the gear were down. He also reduced throttle to idle power.

l/Lt HANKE states that he reduced power to approximately 90%. Separation between aircraft commenced immediately even though the wingman had a power advantage. Although the pilot was not aware of a flame-out it is therefore reasonable to conclude that it occurred at this time. The HIO states that the pilot cycled his gear at 250 KTS as separation increased. It is more probable that the pilot continued to cycle his gear until he sensed the flameout which he states was in the vicinity of 280 KTS, somewhat earlier than the HIO's estimate.

At this first indication of power loss, the pilot made a visual check of the MPM & EGT, and confirmed loss of engine power. He then extended the Ram Air Turbine, advanced the throttles to military power, selether transfer switch to external tanks, depressed the ignition buttons and

ordered the RIO to eject. He visuall checked his engine master switches on, and manually verified that the generator switches were on. There was no recollection of fuel flow, nor did he see a bus tie or generator warning light.

The pilot feels that the generators went off the line at a speed in excess of 250 KTS and probably prohibited the MIO from hearing the command to eject. The RIO was again in disagreement with the pilot on airspeed, estimating ICS loss to be at 240 KIAS.

Apparently no electric power was received from the RAT after the extension.

When the ICS went off the line the pilot utilized the NATOPS hand signal for RIO ejection. The RIO did not see this motion.

At this point the pilot focused his attention on selecting a crash site clear of populated areas. He finally spotted a clearing, steered the A/O towards it and ejected at approximately 800 ft MSL and 160 KIAS. The A/C felt sluggish during this last maneuver but handled well up to that time.

The RIO had not been surprised by the loss of ICS due to previous habit patterns developed between himself and the pilot. When his gyro tumbled he tested his emergency ICS and missed the pilots signal to eject. He did notice the pilot reaching for his face curtain, and ejected immediately following the pilots exit. The airspeed and altitude were somewhat lower and the nose of the airplane was pitching down.

Although the pilot was bent forward and sustained back injuries, both members of the crew made successful egress from the airplane. All items of the escape system functioned perfeatly.

The plane crashed in a wooded area midway between the landing points of the pilot and RiO. The crash site was estimated at 6 miles on the 190 degree radial of Atsugi Tacan.

The pilot and RIO were picked up by helicopter and returned to NAS Atsugi hospital for examination.

# Part VI - Damage To The Aircraft

MiB BuNo 151460 sustained ALFA damage upon collision with the ground in uncontrolled flight. Flight path angle was about 60° below the horizontal; angle of impact was at approximately 40° on a 20° the horizontal; angle at impact was approximately 10° right wing down, slope. Bank angle at impact was approximately 10° right wing down, airspeed at ground contact was low, probably between 170 and 200 knots. Airspeed at ground contact was low, probably between 170 and 200 knots. Angine RFM was low, probably windmill; engine damage was restricted to impact damage in the first two stages and post-crash fire damage, to impact damage in the first two stages and post-crash fire damage, there was no apparent rotational damage. The landing gear were down and locked at the time of the crash. As the pilot ejected, the airard pitched forward and accelerated slightly before impact. The aircraft struck and clipped off several small pine trees just prior to ground impact; damage from these trees was negligible.

The fuselage forward of station 303.62 was reduced to shrapnel upon initial impact. The shrapnel was buried in the hole dug by the nose or sprayed throughout the crash site. Greatest travel of thrown pieces was about 80 yards.

At impact, the wings tore off at the fuselage mold lines and twisted forward to the inverted position. Still inverted, they came to rest 10 yards forward of the impact point.

The fuselage aft of station 303.62 separated up and forward on initial impact and came to rest, pointing opposite to direction of travel, 50 yards forward of the impact point. The engines and aft travel, 50 yards forward of the impact point. The engines and aft travelage were essentially intact, although extensively damaged by the burning of fuel from fuselage cells 3, 4, 5 and 6.

Fuel system components, engines, and engine accessories, generators and their constant speed drives have been transferred to O&R, NAS North Island, San Diego, California for priority DIR.

VMFA-531 ser 2-64A, 31 July 1964, FLB Buno 151460, pilot HANKE

Part VII - The Investigation and Analysis (b) (5)

Special handling required in accordance with para: 66 OPMAVINST. P3750.68

WF 1.-531 ser 2-644, 31 July 1964, F4B BuNo 151460, pilot HANKE

Special handling required in accordance with para, 66 OPNAVINST P375046E

VMFA-531 ser 2-644 31 July 1964, F4B BuNo 151460, pilot HANKE (5) (5)

Special handling required in accordance with

VMFA-531 ser 2-644 31 July 1964, FLB BuNo 151460, pilot HUNKE

(b) (5)

Special handling required in accordance with para. 66 OPNAVINST P3750.62

WF1.-531 ser 2-644 31 July 1964, F4B Buno 151460, pilot HANKE

VMFA-531 ser 2-64A 31 July 1964, F4B Buno 151h60, pilot HANKE

VMF -531 ser 2-644 31 July 1964, F4B BuNo 151460, palot HANKE

Special handling required in accordance with

WFA-531 ser 2-64A 31 July 1964, FLB Bune 151460, pilot MANKE Part VIII Conclusions

Special handling required in accordance with para. 66 OPNA VINST P3750.6E

MFA 532 SET 2-64A 31 July 1964, F4B Runo 151460, pilot HANKE

IX. Recommendations.

(b) (5)

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Provide hundling required in accordance

# All statements withheld under exemption (b)(5).

# VMFA-531 ser 2-64A 31 July 1964, F4B BuNo 151460, pilot HANKE

SGO: NO. LAWRENCE GDO: CAPT HOUSE DUTY SECTION: 3

MARINE FIGHTER/ATTACK SQUADRON-531

Marine Aircraft Group-11

lst Marine Aircraft Wing, Aircraft, MF, Pacific c/o Fleet Post Office, San Prancisco, California, 96601

FLIGHT SCHEDULE FOR

FRIDAY

31 July 1964

CDO: 0800-1230 LT DOISES 1230-1700 CAPT TEMPER 1700-886, CMO SCHMARE

EVENT	PILOT/RIO	ETD	BLOCK TIME	MISSION	TANK		
1-2	MAJ SMITH/STRAYHORN DOWLING/COX	0900	1030 3035	OCI	1300	3/00	NOTES: 1. Lead crew brief hop 1 hour prior to scheduled launch.
2-2	HRUINOME / HERSON HAMER/SCHMARZ	1200	1330	OCT	1400	3/00	prior to ETD.
3-1 3-2	IF COL MOGRAM/SCROUGHS	1500	1630 1635	ect	1700	3/00	3. Requirements as set forth in
12	MAJ METERO/PRAZTER GRESS/CIDATINIS	1800	1930	GOI	2000	260	Paregraph 2016s of Mas Ataugi, Air Operations Marmal have been complied with.

D. T. DOISEE S-3 ADMIN. OFFICE

PARA. 66 Special handling required in accordance with OPMAV INST. 3750.65

CERTIFIED A TRUE CONY.

# U.S. NAVAL AIR STATION ATSUGI, JAPAN METEROLOGY DIVISION

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VMFA-531 ser 2-64A 31 July 1964, FuB Buno 151460, pilot HANKE

FACILITY: YOKOTA APPROACH CONTROL

SUBJECT: THANSCRIPT OF ALL THANSMISSIONS MADE ON APPROACH CONTROL FREQUENCY 343-47MCS

DATE: 31 JULY 1964, 0305Z - 0314Z

ABBREVIATIONS: 2-1 - CHIEFTAIN 2-1

2-2 - CHIEFTAIN 2-2

Y - YOKOTA DEPARTURE CONTROL

TIME- LOCAL TIME - INDIA

TIME	ACFT/FCLTY	TRANSMISSIONS
1205	2-2	Yokota Approach Control, Chieftain 2-2 off the deck at this time climbing VFR.
	Y	Roger Chieftain, report on top.
1	2-1	Check your gear (garbled) your gear is down.
1206 *	2-1	2-2, 2-1 How do you read?
	Y .	Chieftain 2-1, Yokota.
	2-1	Roger 2-1, go.
	Y	2-1, roger, report on top.
	2-1	Roger, my playmate can't get his gear up. I'm slowin, down considerably waiting for him to get them up.
	Y.	2-1, roger.
	2-1	2-2, 2-1 how do you read?
1207	Y	Chieftain 2-2 Yokota Departure, how do you hear?
	<b>Y</b>	Chieftain 2-2, Chieftain 2-2, Yokota transmitting on guard if you hear squawk Mode 3 Code 71 if you are at safe altitude.
1508	Y	Chieftain 2-1 Yokota.
	Y	Chieftain 2-1 Yokota Departure.
	. I	Chieftain 2-1 Yokota Departure.

Y

Chieftain 2-1 Yokota Departure, how do you hear? (Aircraft answered on other channel).....Roger, Request you come back 34 34

1209

Chieftain 2-1 is back up.

SEVERAL TRANSMISSIONS RELATED TO SAR ACTIVITIES ARE CMITTED.

AS CUSTODIAN OF THE CRIGINAL AECOADING, I HEARBY CEATIFY THIS TO BE A TURE AND EXACT TRANSCRIPT THEREOF.

/s/ CHA. LES W JOHNS CHARLES W JUHNS Major, USAF Chief Controller

CERTIFIED A TRUE COPY

para. 66 OPMA VINST. P3750.6E

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DIAGRAM of INCIDENT SHOWING WIND, DIRECTION, APPL CH OF EQUIPMENT, POSITION OF AIRCRAFT, DISTANCES, ETC. (Sape and photographe should be included, if significant) d photos show location of roads in reference to the erash 

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HANDLING REQUIRED IN ACCORDANCE WITH OPNAYINS 750, RE INSTRUCTIONS: SEE REVERSE

RESCUE REPORT OPNAY FORM 3750-12 (3-63)

BPE

FULL DESCRIPTION OF FIREFIGHTING OR PROTECT ON AT INCIDENT to the arrival of the stack equipment.

At 1210 on 31 July 1964 the Greek phone was actuated; and word was passed that there was a possible off station cresh south of the field. This was scarffirmed a season of with four men departed the station. The Seath Chief and the men followed by some at about 1215, and found the station. The Seation Lander errived at the won extragalabed, and found the area inscensible to vehicles. The fire had seen to the arrival of the parameter had been returned to the station via halo STATEMENT OF PIRE. LIBUTENANT G. F. R. HANKE (D) (6) 7307 USMC concerning VMFA-531 ser 2-644 31 July 1964 F4B BuNe 151460 Pilot HANKS

Statement of Rescue from Pilot of Chieftein 2-2

Special handling required in accordance with para 66 OFMAVINST P3750.65

ENCLOSURE (13)

(b) (5)

G. F. R. HANKE

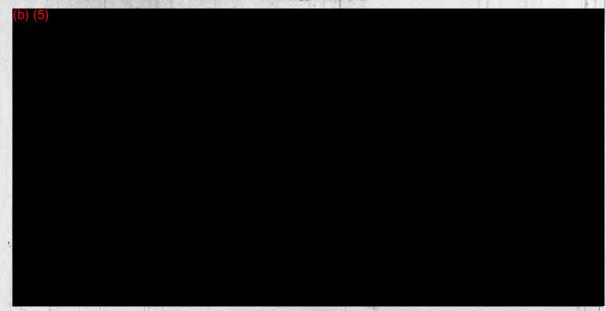
Special handling required in accordance with pera 66 OPNAVINST P3750.68

ENCLOSURE (13)

FROM	50-13 (3-63)			REVERSE	2. DATE OF M	II SHAP	OPMAY DA CA	TE OF RESCUE
00, 1	MFA-531, MAG-11				31 Jul		Control of the Contro	Jul 1964
124 2012 200 200 200 200	Atsugi (SAR)				UH2B	with (Type/s	odel)	
NUMBER OF	SA. IN RESCUE VENICLE OR	1 50. TO BE RESCUED	sc. Rescuts		s. RESCUE NA	CE UP MEANS		
" Smel	3	2	1		SH3A (			
T 15	ME SEQUENCE OF EVENTS	(Local Date Time Gro	up)	84. WATER TO		NDITIONS	The second second second	E SITE
12081	Crash Phone (T	ower Alert)		N		85	•F	1900/10
	Distance to Scene			88. SEA STA	TE/MANE HEIGH	the street of the street of the street	TERRAIN DE	
1214	6 N.M.							
1218	None			11411	y, abou	+ 150 4	4 Mg1	
The second second second	or Bethod of Securing			1 111	J, 2000	150 1	u. no.	
240	Crewman on foo	t					7	
	That Par Sighted First			1. EQUIPMEN	TS ACTUALLY NO	SED DURING HE	scire	
240	Crash scene							
1244	Returned NAS A	tsugi						
S. Survivor(s) Discabarked	Location (If different from	Itee 3)		NA NA				
250	NAS Atsugi Hel:		HE VIEW					
	POST-RETRIEVAL	culties and effect on fin	el netices of	etrer ettespt.	I ALERT	ING PERIOD,	SEARCH/L	LOCATING.
		ched crewman o	n foot to	o conduct	RIO ba	ck to h	elico	pter.
THE RESERVE AND ADDRESS OF THE PARTY OF THE	L REQUIRING RESCUE	GIVE REASON FOR	,		ACTORS COM	PL ICAT ING	RESCUE /	ATTEMPT
ME-LAST	L REQUIRING RESCUE	GIVE REASON FOR	P RESCUE	Physical es	ACTORS COM	PL ICAT ING	RESCUE /	ATTEMPT
ME-LAST	L REQUIRING RESCUE		P RESCUE		ACTORS COM	PL ICAT ING	RESCUE /	
SCHWARZ,	Frank H.	GIVE REASON FOR	P RESCUE	Physical es	ACTORS COM	PL ICAT ING	RESCUE /	ATTEMPT
CHWARZ,	Frank H.	GIVE REASON FOR	P RESCUE	Physical es	ACTORS COM	PL ICAT ING	RESCUE /	ATTEMPT
CHWARZ,	Frank H.	GIVE REASON FOR	P RESCUE	Physical es	ACTORS COM	PL ICAT ING	RESCUE /	ATTEMPT
CHWARZ,	Frank H.	GIVE REASON FOR	P RESCUE	Physical es	ACTORS COM	PL ICAT ING	RESCUE /	ATTEMPT
CHWARZ,	Frank H.	GIVE REASON FOR	RESCUE On	Physical co	ACTORS COV	PLICATING nerence of	RESCUE I	ATTEMPT
CWO USMC	Frank H.	GIVE REASON FOR	SESCUE On	Physical of None	ACTORS COV	PLICATING nerence of	RESCUE I	ATTEMPT
CWO USMC	Prank H.	GIVE REASON FOR	SESCUE On	Physical of None	ACTORS COV	PLICATING nerence of	RESCUE I	ATTEMPT
CHMATZ, CWO USMC	Frank H.	GIVE REASON FOR	SESCUE On	Physical of None	ACTORS COV	PLICATING nerence of	RESCUE I	ATTEMPT
CWO USMC	Frank H.	GIVE REASON FOR	SESCUE On	Physical of None	ACTORS COV	PLICATING nerence of	RESCUE I	ATTEMPT
CWO USMC	Frank H.	GIVE REASON FOR	SESCUE On	Physical of None	ACTORS COV	PLICATING nerence of	RESCUE I	ATTEMPT
CWO USMC	Frank H.	GIVE REASON FOR	SESCUE On	Physical of None	ACTORS COV	PLICATING nerence of	RESCUE I	ATTEMPT
CWO USMC	Frank H.	GIVE REASON FOR	SESCUE On	Physical of None	ACTORS COV	PLICATING nerence of	RESCUE I	ATTEMPT
CHMARZ, CWO USMC	Frank H.	GIVE REASON FOR	on sescue	None None	ACTORS COMMITTEEN, LA	PLICATING herence of	RESCUE A AQUIANA	ATTEMPT
CHMARZ, CWO USMC	CLOSURES: Warratives	GIVE REASON FOR	on sescue on mare/rednique. melicopte	None None	ACTORS COMMITTION, LA	PLICATING herence of	RESCUE A AQUIANA	ATTEMPT
CHWARZ, WO USMC  REMARKS (F-  P110  ATTACH EN  Soul and TIS  C. MCGR	CLOSURES: Warratives	GIVE REASON FOR	on sescue on mare/rednique. melicopte	Physical cyling -S	ACTORS COMMITTEEN, LA	PLICATING norance of	RESCUE A AQUIANA	ATTEMPT
ATTACH EN	CLOSURES: Narratives  CLOSURES: Narratives  CLOSURES: Narratives  AND JR., COMMANDI	GIVE REASON FOR	on sescue on mare/rednique. melicopte	Physical cyling -S	persta/techni	PLICATING norance of	RESCUE A AQUIANA	ATTEMPT

STATEMENT OF CHIEF WARRANT OFFIER F. H. SCHWARZ (b) (6) 6730 USMC concerning WFA-531 ser 2-64A 31 July 1964 FLB Bune 151460 Pilot HANKE

Statement of Rescue from RIO of Chieftain 2-2



F. H. SCHWARZ

Certified Spen Copy

# VMFA-531 ser 2-64A, 31 July 1964, F4B Buno 151460, Pilot HANKE PETROLEUM PADDUCTS LABORATORY ANALYSIS REPORT

Product Nomenclature

JET FUEL, GRADE JP-5

Sample Submitted By (Installation)

MAG-11 ATSUGI

anufacture or Supplier of Product

UNK

Sample Taken By

SOT HARRIS

Name And Location Of Laboratory

QUALITY CONTROL SECTION USAPSDT APO 503

HAKOZAKI LABORATORY

Sample No.

64-6490 & 64-6491

Lab. Report No.

2885-64

wac. No.

MIL-J-5624F

Ant. Prod. Sample Represents

UNK

Source of Sample (Truck, Tank &tc.)

64-319 F4B #8, 64-320 F4B #11

Date Sample Taken

UNK

Date Sample Rec'd

2 AUG 64

Date Tests Started

2 AUG 64

te Tests Compl.

5 AUG 64

	TEST AUSULTS		
	64-6490 64-319 FLB #8	64-6491 64-320 FLB #11	
36.0-48.0 Report	42.5 Clear	42,6 Cle ar	
Report LOO Report Report Report S50 Max	51 Straw 354 384 394 421 476 516 98.0	\$1 Straw 352 383 391 417 471 516 98.5 0.5	
	Report Report LOO Report Report Report Report Report	64-319 FLB #8  36.0-48.0 42.5 Report Clear  Report 354 400 384 Report 394 Report 421 Report 421 Report 476 550 Max 516 98.0 1.5 Max 1.0	

Special handling required in accordance with para: 66 OPNAVINST. P3750.6E

ENCLOSURE (21)

# WFA-531 ser 2-64A, 31 July 1964, MAB BuNo 151460, Pilot HANKE

TESTS .	SPEC LIMITS	TEST HESULTS		
		64-6490 64-319 FLB #8	64-6491 64-320 F4B #11	
PLASH POINT OF A.TER ML/LITER S. DIMEN'T MG/LTTER	11:0 Min	140 None 4.8	140 None 5.4	

### USE LIMITS

# 0.005 Max

Product represented by sample submitted does not meet use limits as outlined in CH-1 CINCPAC INST 4020.6, 5 May 60.

/s/ (b) (6) SFC, NCOIC

CERTIFIED A TRUE COPY

(b) (6)



## U. S. NAVAL AIR STATION NORTH ISLAND SAN DIEGO, CALIFORNIA 92135

O-313-1862

29 October 1964 Information directly related to the below reported occurrence. retained and incorporated into

mishap document.

### SPEEDLETTER

From:

Commanding Officer, U. S. Naval Air Station, North Island.

40731102

San Diego, California 92135

To:

Bureau of Naval Weapons Fleet Readiness Representative, Pacific

(FRR-332/FRR-3331)

Subj:

F-4B BUNO 151460 flameout accident, airframe and engines; investigation

Ref:

(a) COMFAIRWESTPAC 050841Z Aug 1964 NOTAL

(b) BWFRRPAC 111732Z Aug 1964 NOTAL

(c) VMFA-531 040303Z Sep 1964 NOTAL

(d) VMFA-531 150535Z Oct 1964 NOTAL

O-313-1862 29 October 1964

Subj: F-4B BUNO 151460 flameout accident, airframe and engines; investigation

(b) (5)

O-313-1862 29 October 1964

Subj: F-4B BUNO 151460 flameout accident, airframe and engines; investigation of

(b) (5), (b) (6)

By direction

O-313-1862 29 October 1964

Subj: F-4B BUNO 151460 flameout accident, airframe and engines investigation of

Copy to: BUWEPS (FWAE-4/FWAE-1411/F-123/RAPP-2) CGAIRFMFPAC CGFIRSTMAW MARAIRGRU-11 VMFA-314 VMFA-531 COMNAVAIRPAC COMFAIRSDIEGO COMFAIRWESTPAC COMFAIRWESTPAC DET CUBI APPR GE EVENDALE BWR STL MCAS CHERPT NAS QUONPT . BWFRRLANT COMNAVAIRLANT NAVAVSAFECEN NORVA

NATSF PHILA

MARINE FIGHTER/ATTACK SQUADRON-531

Harine Aircraft Group-11

1st Marine Aircraft Wing, Aircraft, FMF, Pacific c/o FPO, San Francisco, California 96601

DP:ckm
8 October 1964

SPEEDLETTER

FROM: COMMANDING OFFICER
TO: COMMANDER, U.S. NAVAL SAFETY CENTER

IN REFERENCE TO YOUR LETTER SER: 22/1518, DATED 24 SEPTEMBER 1964 THE ANSWERS TO YOUR THREE QUESTIONS IN ORDER ARE:



W. C. MCGRAW JR.

RR RUCKDG
DE RUCMBW 931 16/2125Z
ZNR
R 162106Z
FM BUWEPSREP STL
TO RUECM/BUWEPS
INFO RUWDAF/COMNAVAIRPAC
RUCKDA/COMNAVAIRLANT
RUCKDG/NAVAVSAFCEN
RUEGFA/ASO PHILA
RUVDAK/FITRON INO ONE
RUWDAK/FITRON ONE TWO ONE
RUWDAF/BWFRRPAC
RUCKDP/BWFRRPAC
RUCKDP/BWFRRLANT
BT

Saguen 6634

2205 UL

UNCLAS F-4 FUEL BOOST PUMPT TEST SWITCH

A. FITRON 21 112506Z SEP (NOTAL)

B. VMFA-531 319621Z JUL (NOTAL) ARR

D. BUWEPS LTR RAME-532/182:DLK DTD 26 NOV 63

1. UPON RECEIPT OF REF A, EWR REQUESTED FROM CONTRACTOR THOROUGH INVESTIGATION AND COMMENTS/RECOMMENDATIONS ON PROBABILITY OF FAILED BOOST PUMP TEST SWITCHES AS A FACTOR IN REF B ACCIDENT AS WELL AS OTHER UNDETERMINED FLAMEOUT OCCURRENCES.

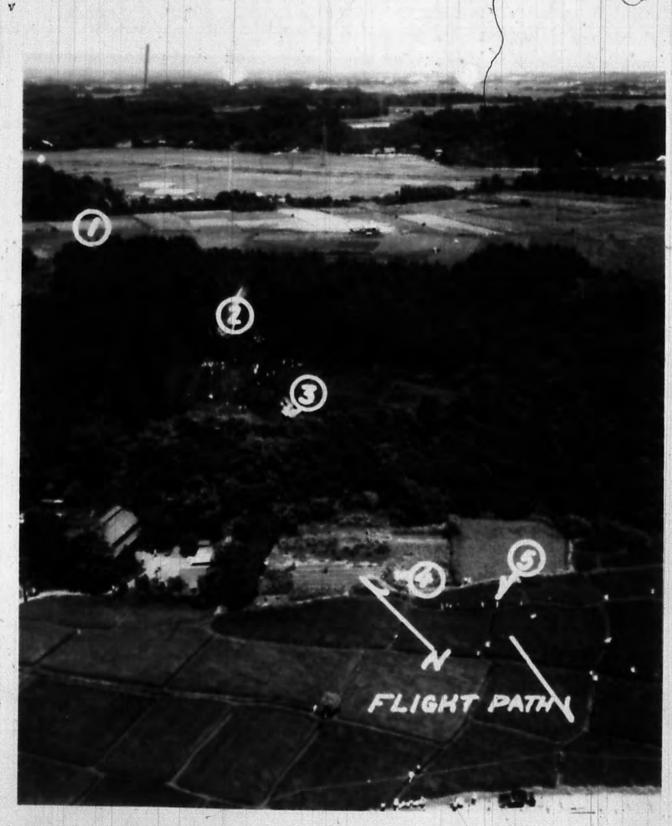
2. CONTRACTOR CURRENTLY CONDUCTING EXTENSIVE INVESTIGATION AND

2205 Uci

PAGE TWO RUCMBW 931 162106Z UNCLAS COMMENTS/RECOMMENDATIONS WILL FOLL ONCE INVESTIGATION COMPLETED. 3., REF C ON SAME SUBJ DISCUSSED AND PASSED TO CONTRACTOR FOR ACTION. 4. CONTRACTOR CONGURS THAT BOOST PUMP TEST SWITCH BURNING CAN OCCUP. PUMP/MAIN FUEL VALVE IS TESTED IMPEDIATELY AFTER FIRST, THEREBY INTERRUPTING FIRST FUEL SHUT OFF VALVE CLOSING ELECTRICAL PWR. NORMAL OPENING/CLOSING TIME OF MOTOR OPE ATED MAIN FUEL VALVE IS APPROX 1 SEC. 5. PARA 5 REF C RECOMMENDS THAT OTH FUEL BOOST PUMP TEST SWITCHES IN ALL F-4 ACFT IN SVC BE REPLACE: WITH NEW MS35059-26 SWITCHES. REF D APPROVED AS A CLASS II CHANGE AN IMPROVED BOOST PIMP TEST SWITCH P/N 25307-26 TO REPLACE THE MS35059-26 SWITCH. PRODUCTION EFFECTIVITY OF NEW SWITCH IS BLOCK 22 AND UP. RETROFIT WAS RECONMENDED TO BE DONE ON AN ATTRITION BASIS. REC. MMEND ASO REVIEW PRESENT STOCK STATUS OF MS 25307-26 SWITCHES. 6. REGARDLESS WHICH SWITCH INSTALLED, BWR RECOMMENDS THAT PILOT BOOST PUMP CHECK PROCEDURE BE ACCOMPLISHED BY WAITING APPROX 2-3 SECS BETWEEN SWITCH ACTUATION TO ALEVIATE CONDITION MENTIONED IN PARACRAPH 4.

> 10 FAAA 7-31-64

F88 15-1460



ENCLOSURE (14) VMFA-531 Ser 2-64A 31 July 1964, F4B, BuNo 151460, Pilot Hanke. 1) Pilot position. 2) Aircraft impact point. 3) Aft fuselage and engines. 4) RIO parachute. 5) RIO seat. Special Handling Required in accordance with paragraph 66 OFNAVINST P3750.6E.



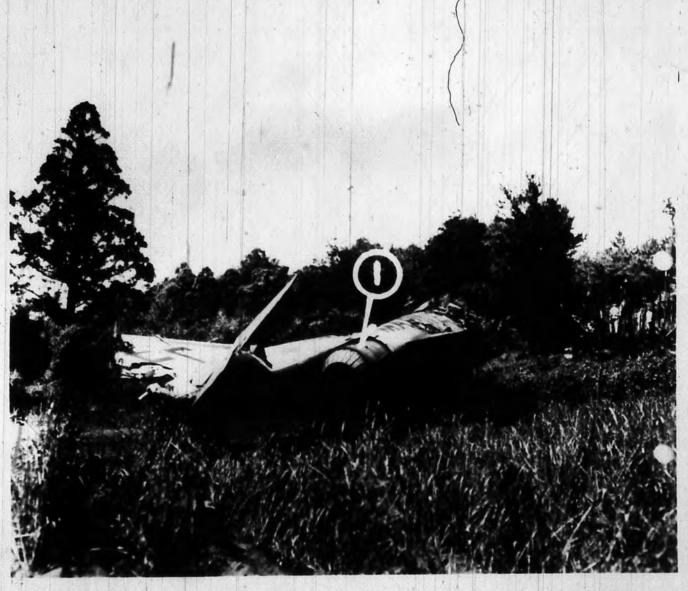
Associate pilot sest position. 2) Aircraft impact point. 3) Starboard wing. 4) Forting to pilot sest position. 2) Aircraft impact point. 3) Starboard wing. 4) Forting. 5) Aft fuselage and engines. Special Handling Required in accordance with eragraph 66 FRAVINST F3750.6E.



ENCLOSURE (18) VMFM-531 Ser 2-64A 31 July 1064, F4B, BuNo 151462, Filot Hanke, showing view along flight path from initial impact point. 1) Fort wing with MIG extended and locked. Special Handling Required in accordance with paragraph 66 OFNAVINGT F0756.6E.



ENCLOSURE (19) VMFA-531 Ser 2-64A 31 July 1964, F4B, BuNo 151460, Pilot Hanke, showing forward view of engines and aft fuselage. 1) Extended ram air turbine. Special Handling Required in accordance with paragraph 66 OPNAVINST P3750.6E.



ENCLOSURE (24) VMFA-531 Ser 2-64A 31 July 1964, F4B, BuNo 151460, Pilot Hanke, showing rear view of engines and aft funciage. 1) Exhaust nouzles closed. Special Handling Required in accordance with paragraph 66 OFNAVINST

The Medical Officer's
Reports were withheld
entirely under
exemptions (b)(5)
and/or (b)(6) of the
FOIA.